

$\ddot{\theta} = \alpha$ $\alpha = \dot{\omega} = \ddot{\theta}$
 $\ddot{\theta} = \alpha$
Linear Momentum

Angular Momentum

$\frac{KE}{\frac{1}{2} m v^2}$
 $\frac{1}{2} I_c \omega^2 = KE$

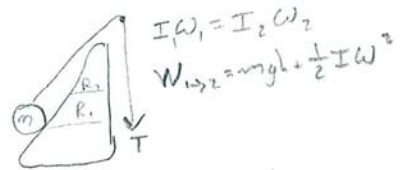
$\sum F = \frac{d}{dt}(m \underline{v})$

$\underline{h}_B = \underline{r} \times m \underline{v}$

$\dot{\underline{h}}_B = \dot{\underline{r}} \times m \underline{v} + \underline{r} \times m \dot{\underline{v}}$

$\underline{\tau}_B = \dot{\underline{h}}_B + \underline{v}_B \times m \underline{v}$

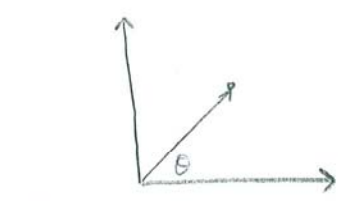
Work Energy Principle



$\underline{H}_B = \underline{H}_c + \underline{r}'_c \times \underline{p}$

$\underline{\tau}_B^{ext} = \dot{\underline{h}}_B + \underline{v}_B \times \underline{p}$
 \downarrow
 $m \dot{\underline{r}}_{cB}$

Do Small & approx @ end



$\hat{e}_r = \cos\theta \hat{i} + \sin\theta \hat{j}$

$\dot{\hat{e}}_r = \dot{\theta} \hat{e}_\theta$

$\hat{e}_\theta = -\sin\theta \hat{i} + \cos\theta \hat{j}$

$\dot{\hat{e}}_\theta = -\dot{\theta} \hat{e}_r$



LMB: Linear Momentum Balance

$\sum F = \frac{d}{dt} m \underline{v}$

$-T \hat{e}_r + mg \hat{i} = m \frac{d}{dt} (l \dot{\hat{e}}_r)$

$= m l \ddot{\theta} \hat{e}_\theta + m l \dot{\theta} \dot{\hat{e}}_\theta$

$-T \hat{e}_r + mg \hat{i} = m l \ddot{\theta} \hat{e}_\theta - m l \dot{\theta}^2 \hat{e}_r$

$-mg \sin\theta = m l \ddot{\theta}$

AMB: Angular momentum balance

$\underline{I}_c = \frac{d}{dt} \underline{h}_c$

$\underline{r}_c \times \sum \underline{E} = \frac{d}{dt} (\underline{r}_c \times m \underline{v})$

$l \hat{e}_r \times (-T \hat{e}_r + mg \hat{i}) = m \frac{d}{dt} (l \dot{\hat{e}}_r + l \dot{\theta} \hat{e}_\theta)$

Energy

$KE + PE = const$

$\frac{1}{2} m v^2 + mgh = const$

$\frac{1}{2} m l^2 \dot{\theta}^2 - mgl \cos\theta = const$

$\underline{v} = l \dot{\theta} \hat{e}_\theta$

$h = -l \cos\theta$

Principle of Virtual Work

$\delta W = \sum_i \underline{F}_i \cdot \delta \underline{r}_i = 0$

$= \sum_j \underline{\Xi}_j \cdot \delta \xi_j$

Lagrange's

$\frac{d}{dt} \left(\frac{\partial L}{\partial \dot{\xi}_j} \right) - \frac{\partial L}{\partial \xi_j} = \underline{\Xi}_j$

step 1 Degrees of Freedom

step 2 conservative / non conservative forces

step 3 $L = T - V$

step 4 $\underline{\Xi}_j$ $F_c \delta c + F_D \delta D + \dots \underline{\Xi}_j \delta \xi_j$ only non-conservative

step 5 complete Lagrange

$V = \frac{1}{2} k x^2 + mah$ $T = \frac{1}{2} m v^2 + \frac{1}{2} I_c \dot{\theta}^2$

$\underline{\Xi}_j \cdot \delta \xi_j$

Process

- Coordinates
- write position vectors
- take derivatives
- $F = ma$

Moments of Inertia

$I_B = I_{cm} + m r_{cm \rightarrow B}^2$

$\frac{1}{2} m R^2$ disk

$m r^2$ point

$\frac{1}{12} m l^2$ rod

$\delta W = \underline{\Xi}_1 \delta x_1 + \underline{\Xi}_2 \delta x_2$

$\underline{v}_B = \underline{v}_A + \underline{\omega} \times \underline{r}_{AB}$

Coefficient of Restitution

$e = \frac{\text{rel vel after}}{\text{rel vel before}}$

$e = \frac{v_2' - v_1'}{v_1 - v_2}$

$e = 0$ masses stick together

$e = 1$ perfectly elastic case KE conserved.

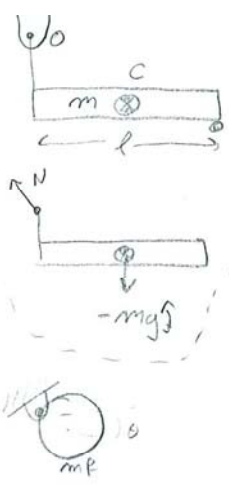
e Only affects direction of impact.

Breaking down ODE's

$X_1 = x$ $\dot{X}_1 = \dot{x}$

$X_2 = \dot{x}_1 = \dot{x}$

$\Rightarrow \dot{X}_2 = \ddot{x}$ & so forth



$$\sum F = m a_{cm} \quad \dot{\omega} = \alpha$$

$$N - mg \hat{j} = m \alpha \times r_{oc}$$

$$= m \ddot{\theta} \hat{k} \times \left(\frac{l}{2} \hat{i} - h \hat{j} \right)$$

LMB $\rightarrow \sum F = m a_c$
 AMB $\rightarrow \sum \tau_o = I_o \alpha$

Trig

$$\sin \theta = \theta - \frac{\theta^3}{3!} + \frac{\theta^5}{5!} - \frac{\theta^7}{7!} + \dots$$

$$\cos \theta = 1 - \frac{\theta^2}{2!} + \frac{\theta^4}{4!} - \frac{\theta^6}{6!} + \dots$$

$$\sin(2\alpha) = 2 \sin \alpha \cos \alpha$$

$$\cos(2\alpha) = 2 \cos^2(\alpha) - 1 = \cos^2(\alpha) - \sin^2(\alpha)$$

$$\sin(\alpha \pm \beta) = \sin \alpha \cos \beta \pm \cos \alpha \sin \beta$$

$$\cos(\alpha \pm \beta) = \cos \alpha \cos \beta \mp \sin \alpha \sin \beta$$

$$\tan(\alpha \pm \beta) = \frac{\tan \alpha \pm \tan \beta}{1 \mp \tan \alpha \tan \beta}$$

Derivatives

S	cos	C	-csc cot
C	-sin	S	sec tan
T	sec^2	C	-csc^2

Vibrations steps

Get equations of motion
 Separate into matrix form.

$$\underbrace{\begin{bmatrix} m & 0 \\ 0 & m \end{bmatrix}}_{\text{mass}} \begin{Bmatrix} \ddot{x}_1 \\ \ddot{x}_2 \end{Bmatrix} + \underbrace{\begin{bmatrix} k_1+k_2 & -k_2 \\ -k_2 & k_1+k_2 \end{bmatrix}}_{\text{stiffness}} \begin{Bmatrix} x_1 \\ x_2 \end{Bmatrix} = \begin{Bmatrix} 0 \\ 0 \end{Bmatrix}$$

$$e^{jz} = \cos z + j \sin z$$

18.03 $ay'' + by' + cy = 0$

if r_1, r_2 real $y(x) = C_1 e^{r_1 x} + C_2 e^{r_2 x}$
 if $r_1 = 0, r_2$ real $y(x) = C_1 + C_2 e^{r_2 x}$
 if $r_1 = r_2$ $y(x) = (C_1 + C_2 x) e^{r_1 x}$

$M \ddot{x} + K x = 0$
 assume solution $x = \begin{Bmatrix} x_1 \\ x_2 \end{Bmatrix} = \begin{Bmatrix} C_1 \\ C_2 \end{Bmatrix} \cos(\omega t - \phi)$

No constant terms @ equilibrium

$$\ddot{x} + A x - B = 0$$

if + wiggles & stable

$$\omega^2 = A$$

$$x(t) = C_1 \cos \omega t + C_2 \sin \omega t$$

if Real part s_1, s_2 negative \rightarrow stable
 if Real part s_1, s_2 positive \rightarrow unstable
 Real part $s_1, s_2 = 0$ neutrally stable.

put back in equation

$$\begin{bmatrix} -m\omega^2 + k_1 + k_2 & -k_2 \\ -k_2 & -m\omega^2 + k_1 + k_2 \end{bmatrix} \begin{Bmatrix} C_1 \\ C_2 \end{Bmatrix} = \begin{Bmatrix} 0 \\ 0 \end{Bmatrix}$$

Equilibrium stuff

- step 1: set all velocities & accelerations to zero
 find conditions for x 's & displacement.
 step 2: Analyze small disturbances.
 $0 \ll 1, x \ll 1 \quad s = s_0 + \frac{\epsilon}{s-1}$
 step 3: Linearize equations of motion for small disturbances.

Det = 0

$$\begin{vmatrix} \alpha & \delta \\ \gamma & \beta \end{vmatrix} = \alpha\beta - \delta\gamma$$

Solve for ω

$m\omega^2 = k_1$ or $m\omega^2 = k_1 + 2k_2$
 multiple frequency solution

$\omega_1, \omega_2, \omega_n$

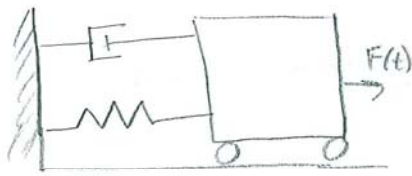
$\omega_1 \Rightarrow \begin{Bmatrix} 1 \\ 1 \end{Bmatrix}$
 $\omega_2 \Rightarrow \begin{Bmatrix} 1 \\ -1 \end{Bmatrix}$

$$\begin{Bmatrix} x_1 \\ x_2 \end{Bmatrix} = A \begin{Bmatrix} 1 \\ 1 \end{Bmatrix} \cos(\omega_1 t - \phi_1) + B \begin{Bmatrix} 1 \\ -1 \end{Bmatrix} \cos(\omega_2 t - \phi_2)$$

A, B, ϕ_1, ϕ_2 the IVP

Gussed

$$\begin{Bmatrix} x_1 \\ x_2 \end{Bmatrix} = \begin{Bmatrix} C_1 \\ C_2 \end{Bmatrix} \cos(\omega t - \phi)$$



$$m\ddot{x} + c\dot{x} + kx = F(t)$$

$X(t) = \underbrace{\text{Free Response}}_{\text{Complementary Solution}} + \underbrace{\text{Response due to Force}}_{\text{particular Solution}}$

$$m\ddot{x} + c\dot{x} = 0$$

$$V = v_0 e^{-ct/m}$$

integrate to get

$$X = x_0 + \frac{mv_0}{c} (1 - e^{-ct/m})$$

if $r_1 = r_2$

$$y(x) = c_1 e^{r_1 x} + c_2 x e^{r_1 x}$$

$$y'' + zy' = 0$$

$$r^2 + zr = r(r+z)$$

$$y(x) = c_1 + c_2 e^{-zx}$$

$$ay'' + by' + cy = 0$$

$$y(x) = c_1 e^{r_1 x} + c_2 e^{r_2 x}$$

18.03

$$m\ddot{x} + c\dot{x} + kx = 0$$

$$\ddot{x} + 2\zeta\omega_n \dot{x} + \omega_n^2 x = 0$$

$$\omega_n^2 = \frac{k}{m} \quad \text{natural frequency}$$

$$\zeta = \frac{c}{2m\omega_n}$$

to solve, assume

$$x = A e^{\lambda t}$$

$$\lambda^2 + 2\zeta\omega_n \lambda + \omega_n^2 = 0$$

$$\lambda = -\zeta\omega_n \pm \omega_n \sqrt{\zeta^2 - 1}$$

$$\omega_d = \omega_n \sqrt{1 - \zeta^2}$$

Periodic function

$$x_p = \text{Re} \{ X e^{i\omega t} \} = \text{Re} \{ |X| e^{-i\phi} e^{i\omega t} \}$$

$$= |X| \cos(\omega t - \phi)$$

$$|X| = \frac{F_0}{k} \frac{1}{\sqrt{(1 - \frac{\omega^2}{\omega_n^2})^2 + (2\zeta \frac{\omega}{\omega_n})^2}}$$

$$\tan \phi = \frac{2\zeta \frac{\omega}{\omega_n}}{1 - \frac{\omega^2}{\omega_n^2}}$$

Overdamped

$\zeta > 1 \Rightarrow \lambda_1, \lambda_2$ Real negative #'s

$$X = A \pm e^{-((\zeta\omega_n + \omega_n \sqrt{\zeta^2 - 1})t)}$$

Critically damped

$$\zeta = 1 \quad \lambda_+, \lambda_- = -\omega_n$$

$$X = (A_1 + A_2 t) e^{-\omega_n t}$$

Under damped

$$0 \leq \zeta < 1$$

$$\lambda_1, \lambda_2 = -\zeta\omega_n \pm i\omega_d$$

$$X = [A_1 e^{i\omega_d t} + A_2 e^{-i\omega_d t}] e^{-\zeta\omega_n t}$$

$$x(t) = C e^{-\zeta\omega_n t} \cos(\omega_d t - \phi)$$

$$C = \frac{A_3}{\cos \phi} \quad \phi = \tan^{-1} \left(\frac{A_4}{A_3} \right)$$

$$A_3 = X_0$$

$$A_4 = \frac{v_0 + \zeta\omega_n X_0}{\omega_d}$$

PS3Q1 Impulse

$$I = \int_{t_1}^{t_2} \vec{F} dt = M\vec{V}_2 - M\vec{V}_1 = \Delta p$$

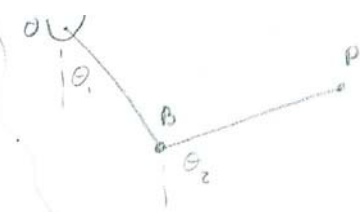
$$I_m = \text{Value} = 10 \text{Ns} \quad m = 1.5 \text{kg} \quad l = 0.6 \text{m}$$

$$\frac{I_m}{2m} = V_2 = 3.33 \text{ m/s} = V_{cm}$$

$$I_{\vec{L}} = \int_{t_1}^{t_2} \vec{r} \times \vec{F} dt = \frac{d}{dt} \int_{t_1}^{t_2} \vec{F} dt = \frac{d}{dt} I_m = 3 \text{ Nms}$$

$$\vec{L} = \frac{d}{dt} (I_{inert} \vec{\omega}) = I_{inert} \omega_2 - I_{inert} \omega_1 = I_{imp} \hat{z}$$

$$\omega_2 = \frac{I_{\vec{L}}}{2 I_{inert}}$$

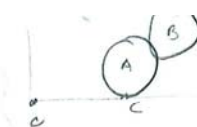


$$y_B = -L_1 \cos \theta_1$$

$$\delta y_B = L_1 \sin(\theta_1) \delta \theta_1$$

$$y_P = y_B + L_2 \sin(\theta_2 - 90)$$

$$\delta y_P = \delta y_B + L_2 \cos(\theta_2 - 90) \delta \theta_2$$



1) Find $v_{0A} = \hat{i} \hat{j}$

$$r_{0B} = \hat{i} \hat{j}$$

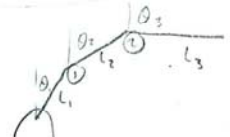
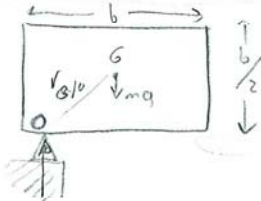
$$\delta r_{0A}, \delta r_{0B}$$

$$V_B = V_A + \omega \times r_{A0}$$

$$\delta r_{0A} = \delta \theta_A \hat{k} + R \hat{j} = \delta x \hat{i}$$

$$\delta \theta_A = \frac{-\delta x}{R}$$

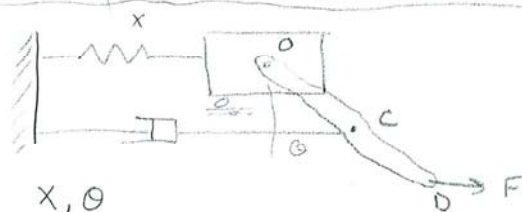
PS5Q3



$$\delta W = \underbrace{FL_1 \cos \theta_1}_{Q_1} \delta \theta_1 + \underbrace{FL_2 \cos \theta_2}_{Q_2} \delta \theta_2$$



DOF: 3
 $\dot{x}_1, \dot{\theta}, \dot{x}_2$
 position vector & take velocity
 do Lagrange's method



$$x, \theta$$

$$V_c = \dot{x}$$

$$r_c = (x + \frac{b}{2} \sin \theta) \hat{i} - \frac{b}{2} \cos \theta \hat{j}$$

$$\dot{r}_c = \dot{x} \hat{i} + (\frac{b}{2}) \dot{\theta} \cos \theta \hat{i} - \dot{x} b \dot{\theta} \sin \theta \hat{j} + \frac{b}{4} \sin^2 \theta \dot{\theta}^2$$

$$V_c^2 = \dot{x}^2 + \frac{b^2}{4} \dot{\theta}^2 + \dot{x} b \dot{\theta} \cos \theta$$

$$\delta r_D = \delta x \hat{i} + \delta (b \sin \theta \hat{i} - b \cos \theta \hat{j})$$

$$F \hat{i} \cdot \delta r_D = F \delta x + F b \cos \theta \delta \theta$$

Similar for r_C

$\delta W = \sum \text{total}$ separate into $\delta \theta, \delta x$
 do La graty

$$\frac{\partial L}{\partial \dot{x}} \rightarrow \frac{d}{dt} \left(\frac{\partial L}{\partial \dot{x}} \right) \text{ all live}$$

$$\frac{\partial L}{\partial x}$$

$$\frac{d}{dt} \left(\frac{\partial L}{\partial \dot{x}} \right) - \frac{\partial L}{\partial x} = \underline{\underline{0}}$$

$$T = \frac{1}{2} m_1 V_1^2 + \frac{1}{2} m_2 V_2^2 + \frac{1}{2} m_3 V_c^2 + \frac{1}{2} I_A \omega_A^2 + \frac{1}{2} I_B \omega_B^2$$

$$V_1 = \dot{x}$$

$$V_A = \dot{x}$$

$$L_{tot} = x_1 + x_2 + y + L_c = \text{const}$$

$$\frac{d}{dt} (L_{tot}) = 0 \quad \dot{x}_1 + \dot{x}_2 + \dot{y} = 0$$

$$\dot{x}_1 = \dot{x}_2$$

$$\dot{x} = -\dot{y}$$

$$\ddot{y} = 2 \ddot{x}$$

$T_n = 0.75$ s period of undamped oscillation

$\frac{x_1}{x_2} = 4$ ratio of two successive positive displacement amp

damping ratio $\zeta =$

viscous damp $C =$

equivalent $K =$

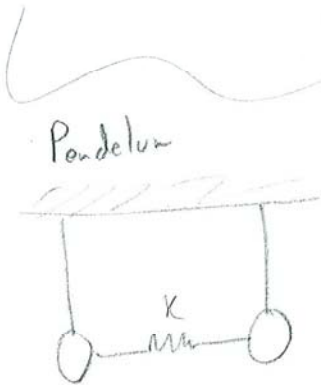
$$\delta = \ln\left(\frac{x_1}{x_2}\right)$$

$$\zeta = \frac{\delta}{\sqrt{(2\pi)^2 + \delta^2}}$$

$$\zeta = \frac{C}{2m\omega_n}$$

$$\omega_n = \frac{2\pi}{T_n}$$

$$\omega_n = \sqrt{\frac{k}{m}}$$



natural frequency

$$\omega_n = \sqrt{\frac{g}{L}}$$

$2x$

$$\omega_n = \sqrt{\frac{g}{L} + \frac{2k}{m}}$$

$$ml^2 \ddot{\theta} + mgL\theta + 2kL^2\theta = 0$$